Since the early 1960s, turbo expanders have been increasingly used in hydrocarbon gas processing plants. e other primary market application of turbo expanders is for the production of industrial gases, such as oxygen.

e production of industrial gases might not typically be recognized as essential to hydrocarbon engineering. However, reducing the production costs of industrial gases is fundamental to the viability of e cient and environmentally responsible global energy solutions, such as oxygen-based synthesis gas production and the liquefaction of natural gas using nitrogen as a refrigerant. e turbo expander e ciency gains achieved in industrial gas turbo expanders can also bene t hydrocarbon gas process plants. As aerodynamic e ciency improvements drive expanders to thinner, longer blades, strength and natural frequency considerations become critical in the design of turboexpander rotors. Reliable use of a higher e ciency turbine design is feasible by restricting speci c ranges of constant machine speed during the startup.

Separating Efficiently

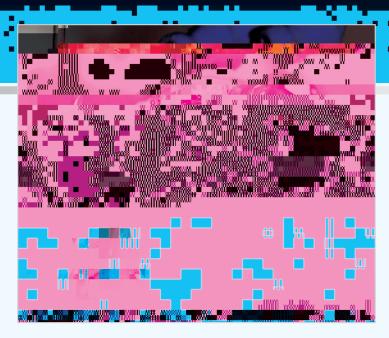
Today's cryogenic air separation plant uses a turboexpander to produce the low-temperature (cryogenic) refrigeration for the fractional distillation and liquefaction of air (-196 °C). A 100% e cient (isentropic) expansion extracts the maximum possible energy from the gas as the pressure is dropped, typically through a radial in- ow turbine. Power extracted from the turbine gas stream usually drives an integral-sha booster compressor to "pre-boost" the turbine. e booster compressor is speci cally sized to load the turbine at the optimum speed at the design process conditions.

Electricity is the chief cost of production in the industry gas industry. e amount of electricity used per Nm3 of gas produced is basically a function of the thermodynamic e ciencies of the compression and expansion of gas. erefore, improving the compressor and turbine e ciencies has been paramount to directly reducing the cost of the product; i.e., the oxygen, nitrogen, and argon gas. Studies commonly reveal that, due to the energy invested to produce cryogenic temperatures, a point increase in turbine e ciency is expanders. e design is a synergy of streamline analysis programs, Computational Fluid Dynamics (CFD), Finite Element Analysis (FEA), 5-axis machining capability, a framework of eld-proven designs, and an invaluable database of cryogenic performance tests. Using these tools, ACD is able to provide the highest e ciency possible while maintaining reliability and unit life.

In addition to evaluating the aerodynamic performance e ects of geometrical modi cations, CFD is an indispensible tool for analyzing the interaction between the inlet guide vanes and the turbine wheel blades. e resulting uctuating pressure eld for a high

The Art Of Design

For over forty years, ACD LLC has strived to continuously improve the e ciency and mechanical reliability of cryogenic turbo



5-axis machined from a single-piece forged aluminum biscuit as shown in Figure 4. Shrouded wheels, however, are a signi cant cost increase and not always required for acceptable operating ranges.

e excitation of potentially damaging wheel natural frequencies and their source of excitation are determined using a Campbell¹ diagram. e vertical axis of the Campbell diagram is for the wheel natural frequency and the horizontal axis is for the rotor speed.

e problematic turbine wheel natural frequencies are plo ed as horizontal lines. Excitation sources are a function of rotational speed, such as multiples of the number of vanes and blades. ese diagonal lines intersect the wheel natural frequencies at the operating speeds that must be avoided. ese intersections are referred to as "interference speeds." A typical Campbell diagram is shown in Figure 5. e excitation sources are multiples of the number of inlet guide vanes (1xV, 2xV, 3xV) and the number of wheel blades (1xB). Dwelling at a constant speed within the two speed ranges, identi ed as "Do-Not-Dwell" zones. will cause resonance and rapid accumulation of damaging fatigue cycles. For example, at a 6,000 Hz natural frequency, a million fatigue cycles accumulate in less than three minutes.

Constant speed operation at interference speeds causes an essentially unbounded ampli cation of the excitation stresses in the turbine wheel. e crystalline structures of metals possess virtually no internal damping to dissipate the excitation energy. When the frequency and shape of the excitation match that of the natural frequency, the condition is referred to as resonance². e more closely the frequency and shape of the excitation directly match that of the natural frequency, the higher the ampli cation factor. For this reason, the lower order natural frequency mode shapes are easily excited by low number multiples of the number of vanes and blades.

Numerous test methods allow veri cation of the natural frequencies and mode shape of a turbine wheel. e simplest and least comprehensive method consists of mounting the wheel on a shaker-table with pickups to measure the response of the blades and wheel. Typically, the response spectrum exhibits the characteristic high ampli cation factor and the narrow frequency band of the undamped natural frequency resonance³.

Ramping Up To Efficiency

A er a considerable amount of engineering and design iterations, an aerodynamically e cient turbine design will likely still have some interference speeds between zero and the trip speed. Fortunately, resonance can be minimized by accelerating or decelerating through these speed zones. For a turbine designed with a conservative speed margin from the design speed, the low inertia rotor allows the turboexpander to quickly pass through the interference speeds to prevent damaging resonance.

If the process requires unrestricted operating speeds from zero to the trip speed, i.e., operation at speeds far from the speci ed design and o -design cases), there are engineering solutions to protect the turbine's mechanical integrity. e wheel's natural frequencies can be increased by thickening the blades and modifying the wheel's disk and hub pro le. is compromise, however, results in a less e cient aerodynamic design.

As the demands on industrial gas turbo expanders rise, cryogenic equipment manufacturers must meet the design challenge. Aerodynamic e ciency improvements can be mechanically reliable if careful study of excitation and natural frequency is performed. Observance of "Do-Not-Dwell" zones o ers additional room for turboexpander advances.

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